



# Gas jet Development at LBNL

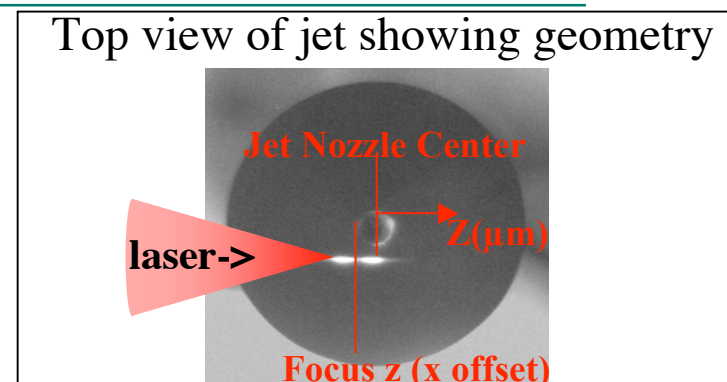
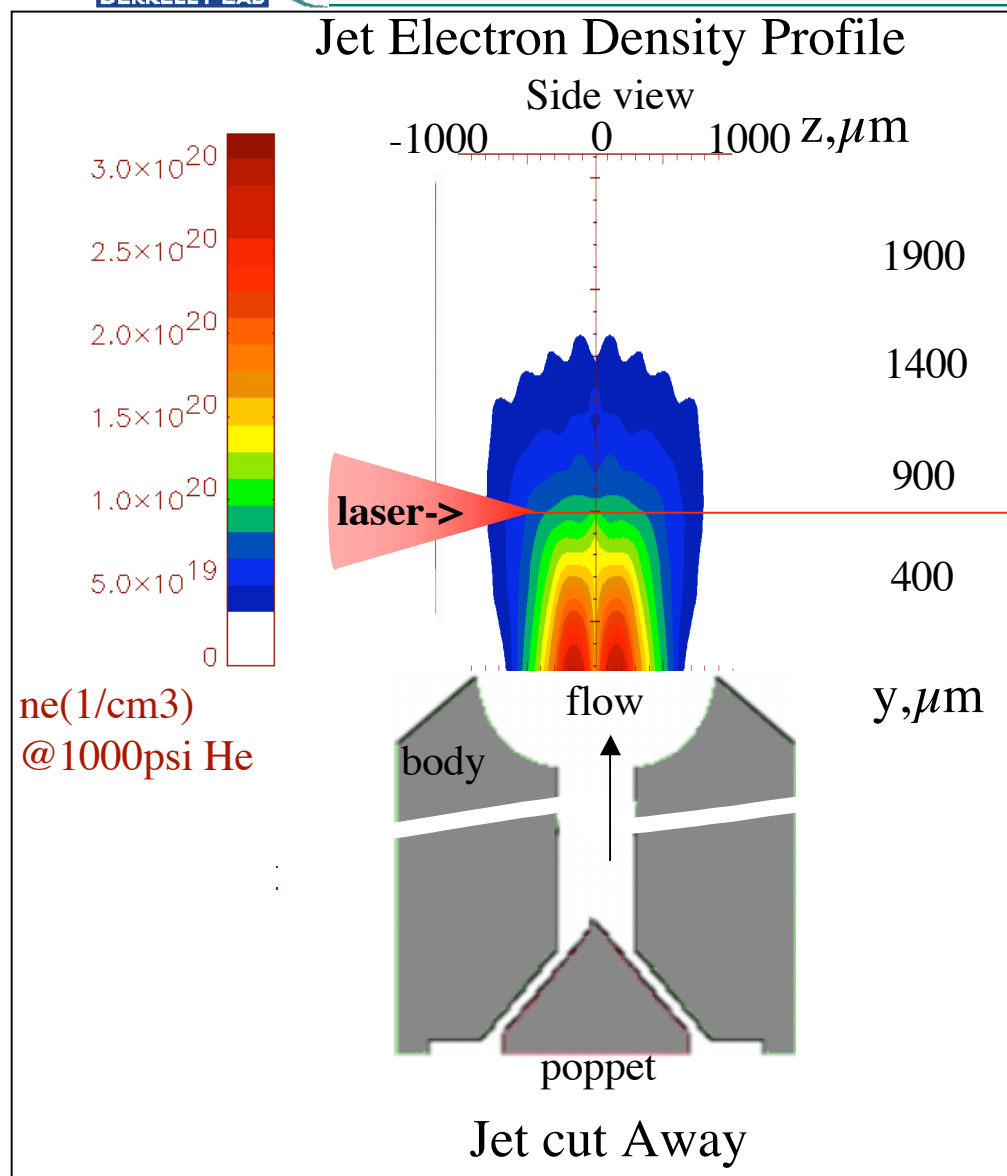
Design, Production, & Test are advancing gas target performance

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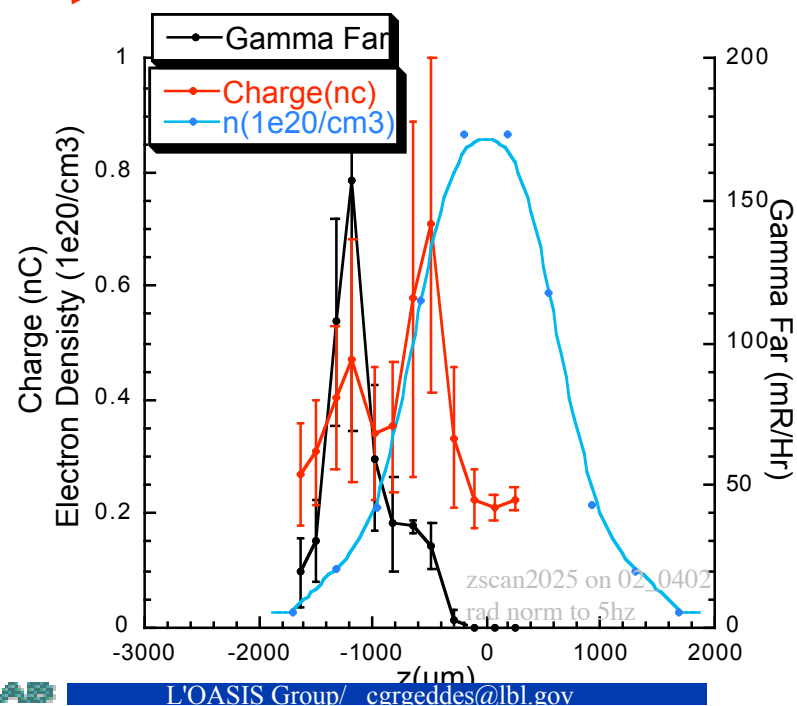
- Gas jets are used in wake field experiments for repeatable, high rep rate plasmas
- Accelerator performance depends critically on jet density, profile, and smoothness as well as on laser focus location in the jet. Jet optimization is likely to significantly improve performance.
- Anticipated needs for optimal performance are challenging to achieve
- Jet development efforts at LBNL include:
  - design and testing of new nozzle shapes (cylindrical and asymmetric)  
fast quasi 1d tools and 3d flow simulations
  - advanced machining to allow production of novel nozzle shapes
  - development of new valve technologies (PZT, micro valves)
  - Test stand for quick characterization of nozzles
  - 2w asymmetric imaging interferometer in the accelerator chamber



Accelerator performance is strongly affected by jet properties including back pressure, profile, z location

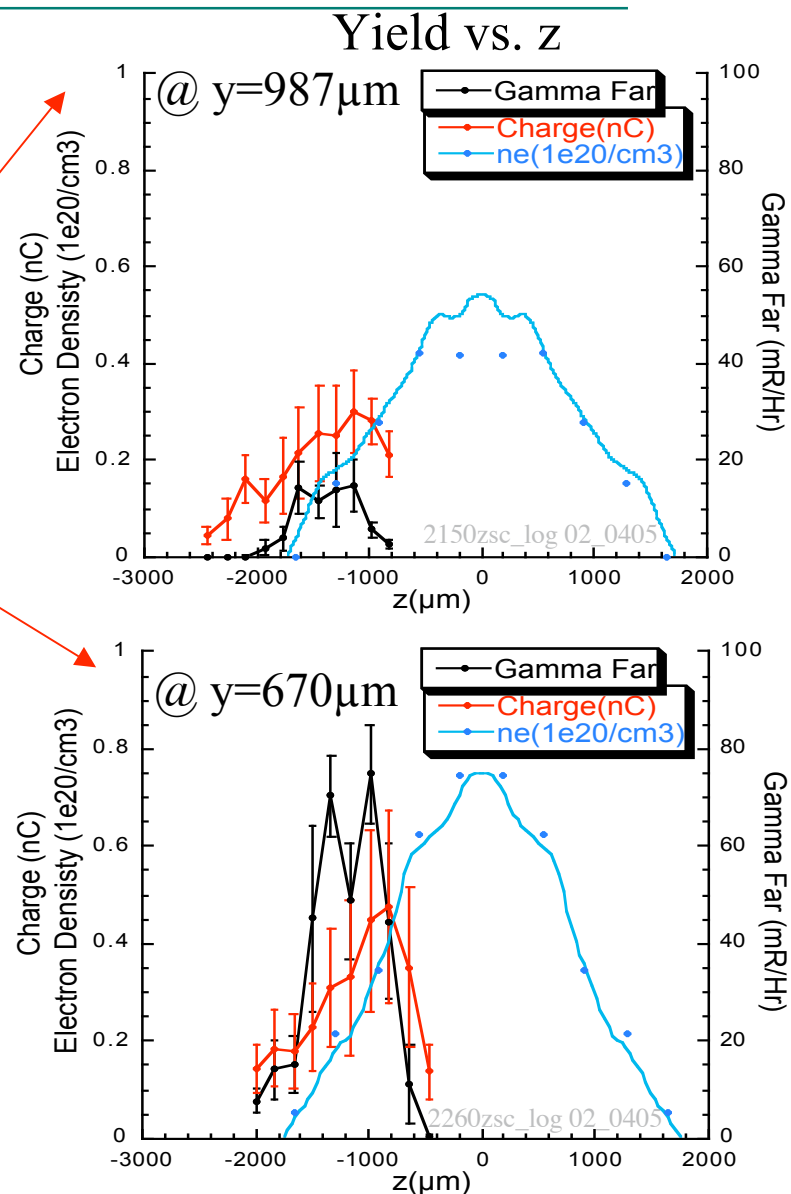
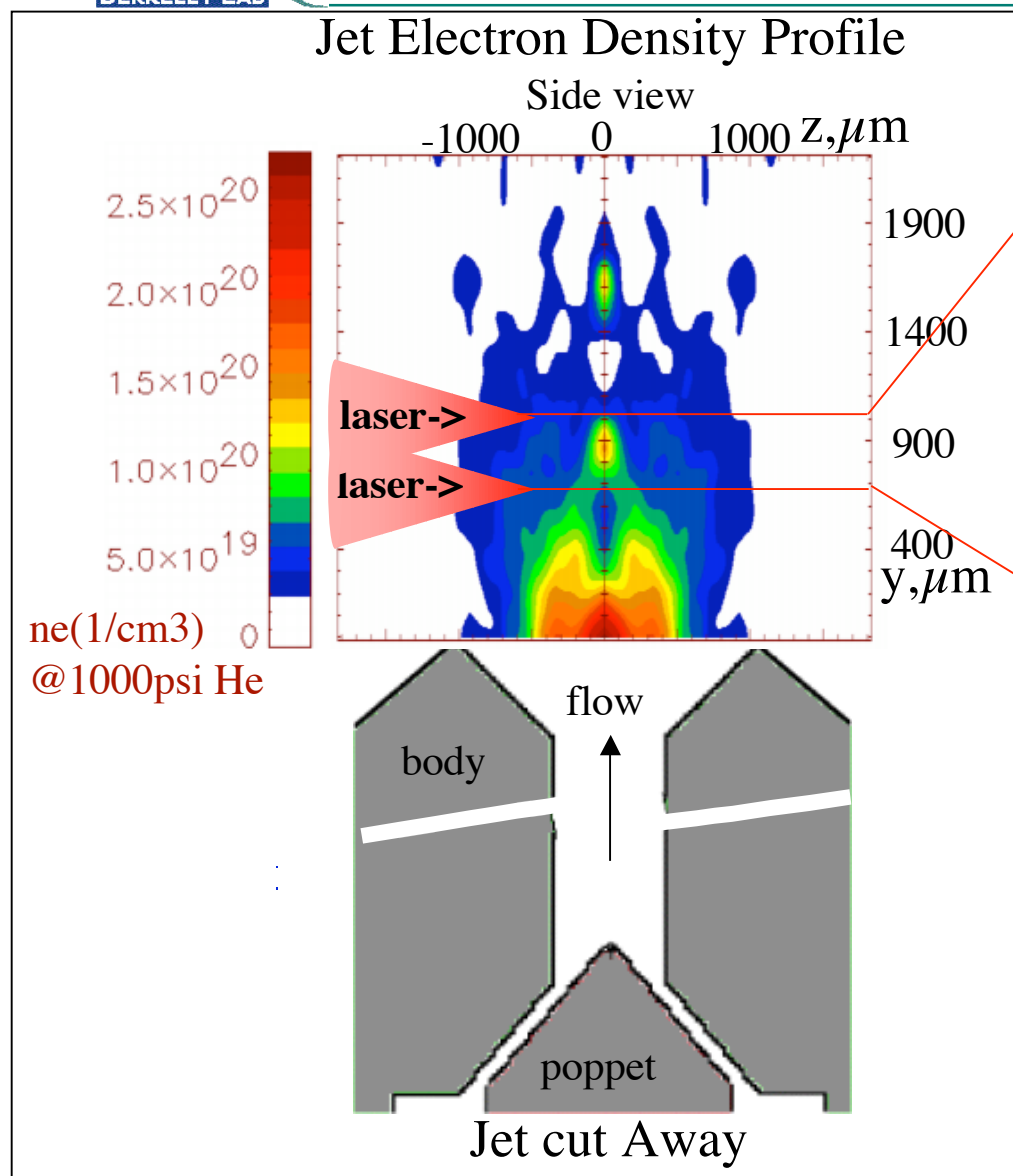


Yield vs.  
focus location, z @ y=710μm





# Jet optimization is under way to control and enhance accelerator output

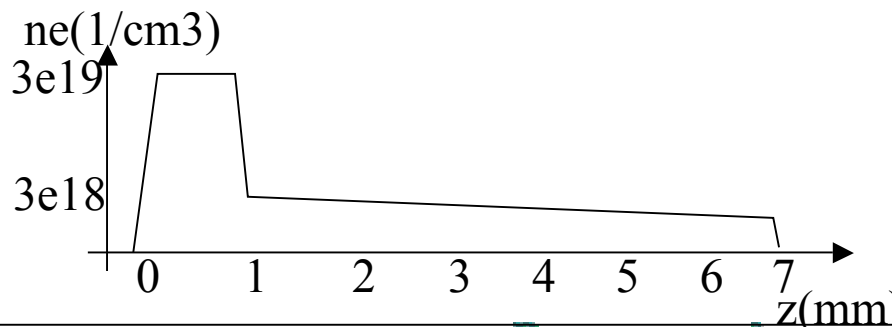




# Wake field experiment gas jet needs & status

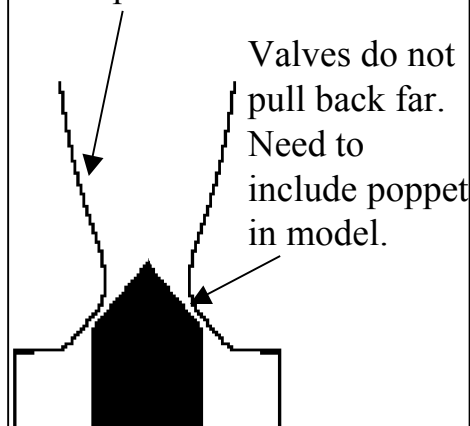

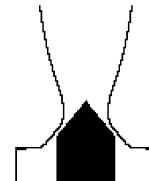
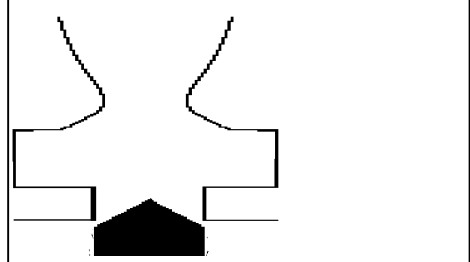
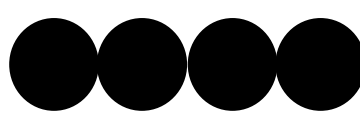
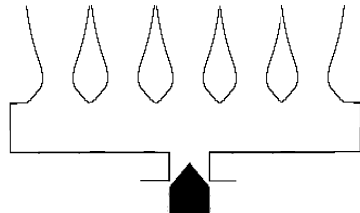


	<u>Desired:</u>	<u>Have</u>	<u>Solutions?</u>
Electron densities (He jet):	1e18 to 1e20/cm <sup>3</sup>	1e18 to 1e20/cm <sup>3</sup>	
Density gradients: (@1mm from nozzle)	~100μm	1mm	highly supersonic
Density fluctuations(@center)	< 5%	?	
2mm diameter round jets	X	X	
10-20mm long rectangular jets (<2mm wide to reduce pumping needs)	X		3d modeling
Fast gas pulses (100μs - 1ms) (practical pumping)	X		new valve technology (piezo, etc under study)
Shaped density profiles (control of injection and acceleration)	X		Multiple nozzles

Sketch of a gas profile of interest as a function of z  
transverse profile insensitive at > 100μm scale





# Sample of a few potentially interesting nozzle shapes

Cut away:	Type	Top View	Side View
<p>Bell or expanding profile for supersonic exit</p>  <p>gas reservoir</p>	Single round nozzle		
<p>A chamber after the poppet may be needed</p> 	multiple round nozzles (fed from 1 valve or several small valves)		
	rectangular nozzle (must fan out more in z than x)		
	?		
	Other ways		



# 1d wall following perfect flow code for fast nozzle modeling of cylindrical & rectangular sections

Use the equations of perfect gas flow in a variable area duct derived from assuming isentropic expansion of a perfect gas):

$$T/t_0 = 1 / ((1 + ((\gamma - 1) / 2) * M^2))$$

$$P/p_0 = 1 / ((1 + ((\gamma - 1) / 2) * M^2)^{(\gamma / (\gamma - 1))})$$

$$\rho / \rho_0 = 1 / ((1 + ((\gamma - 1) / 2) * M^2)^{(1 / (\gamma - 1))})$$

$$A / A^* = (1 / M) * ((2 / (1 + \gamma)) * (1 + ((\gamma - 1) / 2) * M^2))^{((\gamma + 1) / (2 * (\gamma - 1)))}$$

$$M = \text{mach \#}$$

Calculate flow direction at each z location by looking for the minimum length line satisfying (assuming linear variation of flow angle over cross section- first approximation) :

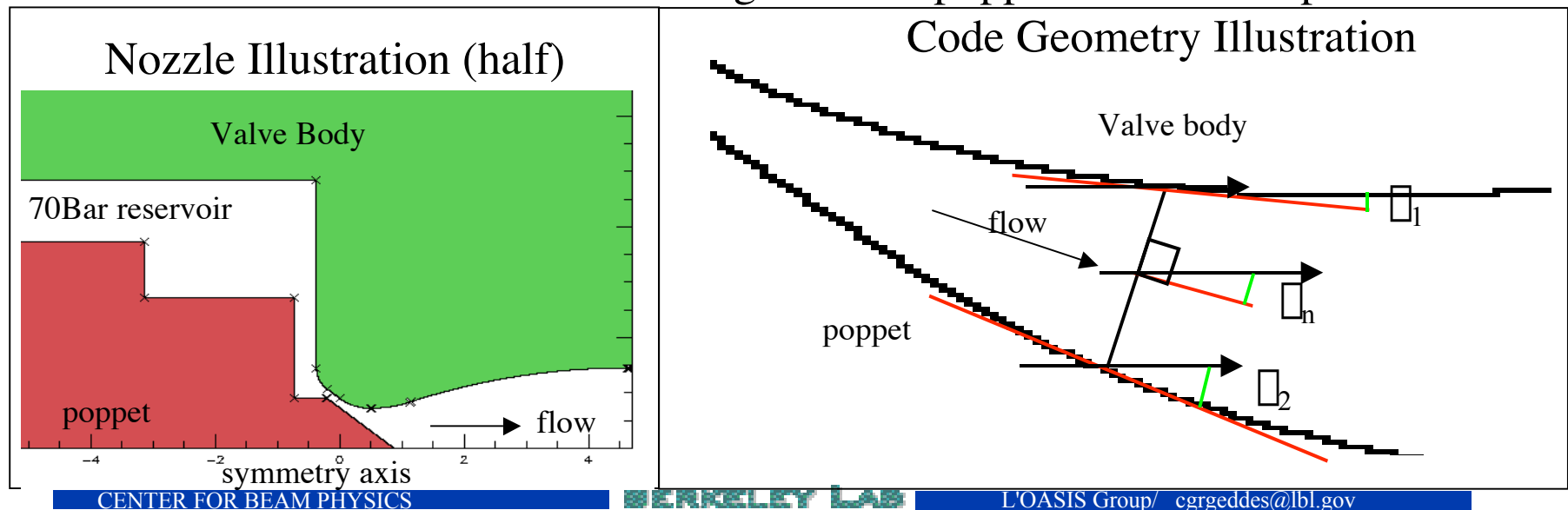
$$\theta_{\text{tan}} = \theta_2 + (b - r_2) (\theta_1 - \theta_2) / (r_1 - r_2)$$

where b is the radius where equal area is inside and outside of r

$$b = \sqrt{(r_2^2 + r_1^2) / 2}$$

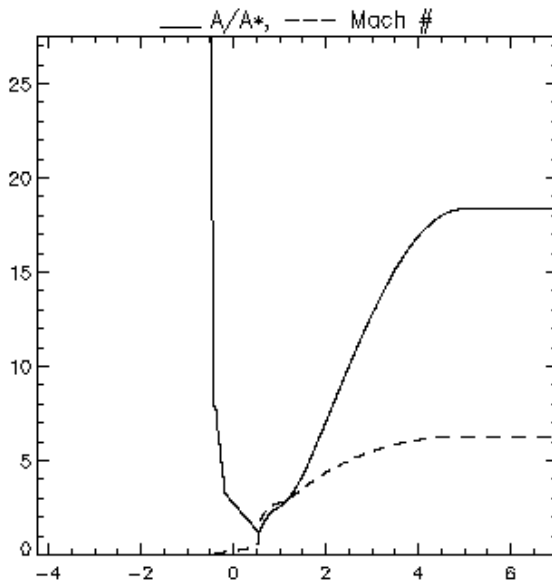
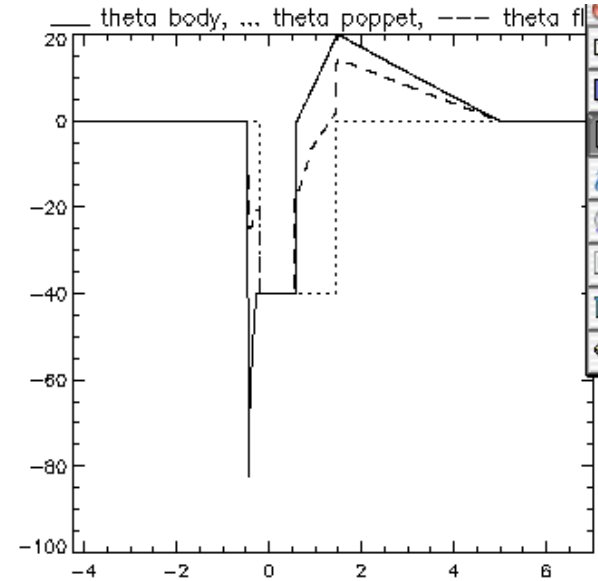
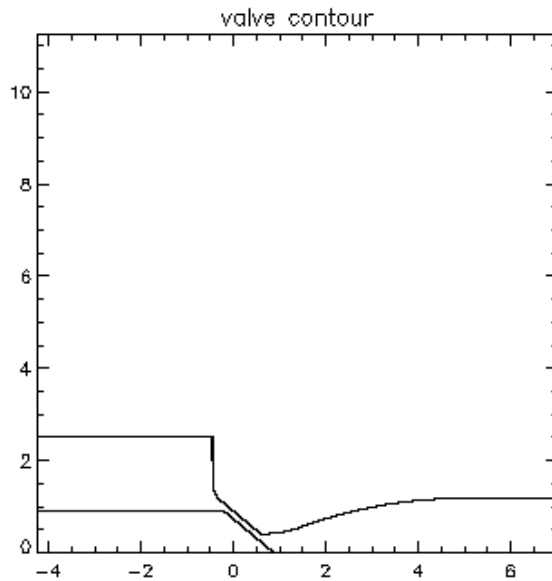
Allows fast optimization of nozzle shape for smooth mach # contour

Includes effects of flow direction changes due to poppet & throat shape.





# Unoptimized nozzles have sharp transitions in Mach # creating shocks (04- GV Bell Nozzle)



;04- GV bell nozzle  
Mon Aug 27 15:25:36 2001

Throat Data:  
At=0.238461, Zt=0.552363

Opening & poppet data:  
opening distance=0.196903, z(open)=0.349833  
poppet area=1.15708, flow area at opening=0.416697  
M at opening=0.350010, Max pressure=172.849

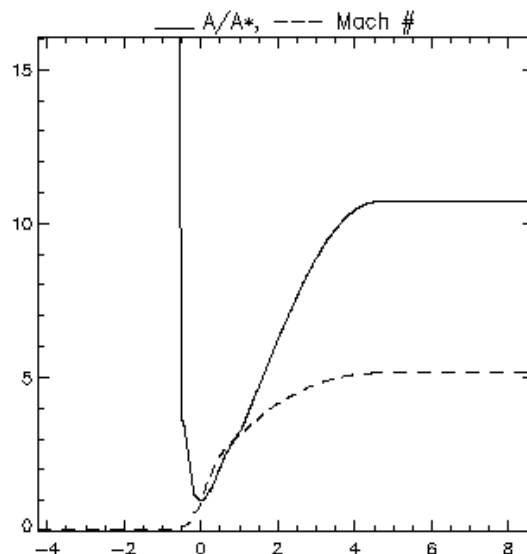
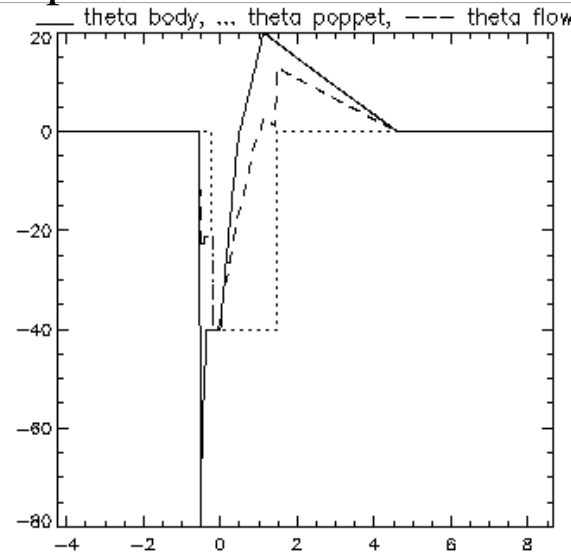
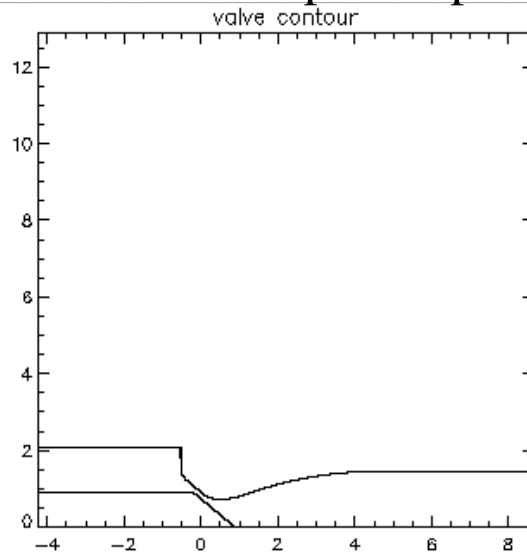
Outflow Data:  
Ae/A\*=18.3364 Ae=4.37253, Re=1.17977  
Me=6.28001, RhomaxE=1.70257e+20, RhoE/rho0=0.0182408

.7e20 rhoe @ 69bar



# 1d Code allows quick nozzle optimization, and output to 3d flow simulations (SciDAC collaboration)

## Sample output - 1d optimized nozzle



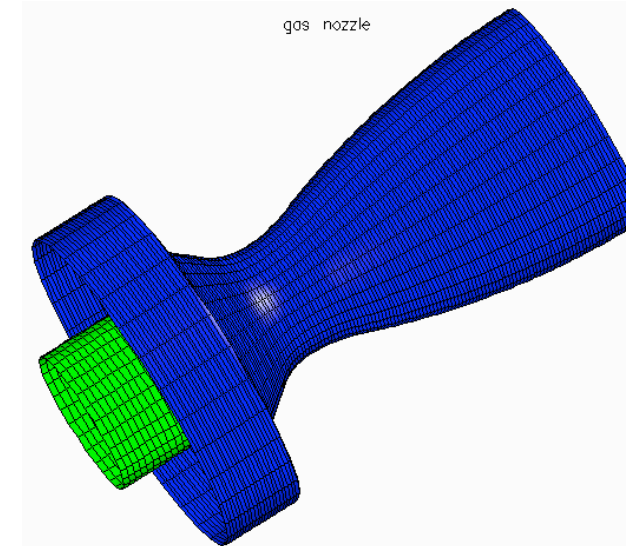
;06- GV bell nozzle redesign  
Mon Aug 27 17:20:26 2001

Throat Data:  
 $A_t=0.619303$ ,  $Z_t=-0.00384712$

Opening & poppet data:  
opening distance=0.193507,  $z(\text{open})=-0.00384712$   
poppet area=2.56647, flow area at opening=0.619303  
 $M$  at opening=1.00001, Max pressure=77.9281

Outflow Data:  
 $A_e/A^*=10.7017$ ,  $A_e=6.62758$ ,  $Re=1.45248$   
 $Me=5.14001$ ,  $Rho_{max}E=1.33768e+20$ ,  $RhoE/rho0=0.0317880$

## Geometry export example



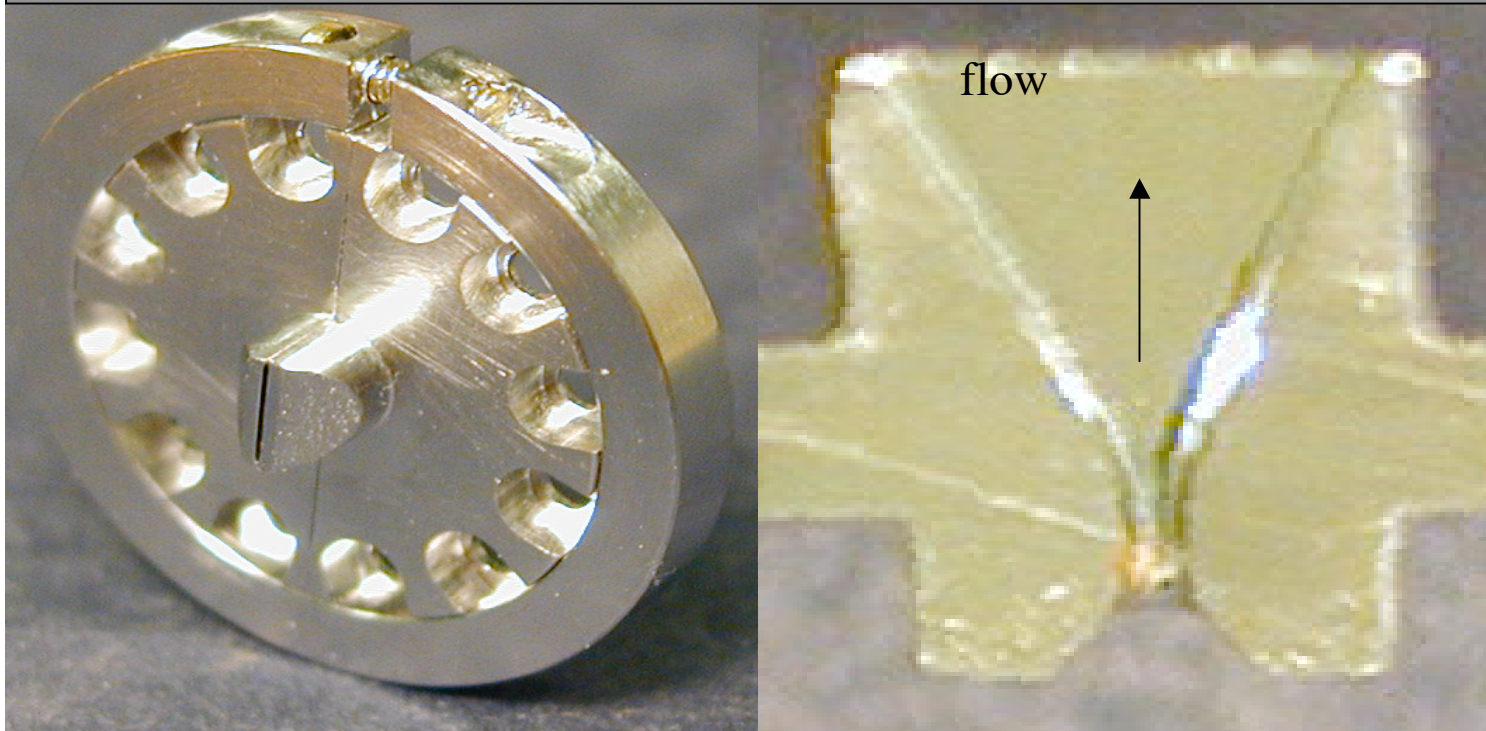
- Geometry export interfaces to 3d flow codes and machining tools
- SciDAC collaboration for 3d flow simulations is under way



## Advanced nozzle shapes are possible through geometry export to CMC machining tools

- Slit nozzles and other shapes not possible with traditional machining are being built using computer machine control with geometries exported from the design tools
- First nozzle has been cut and testing is beginning

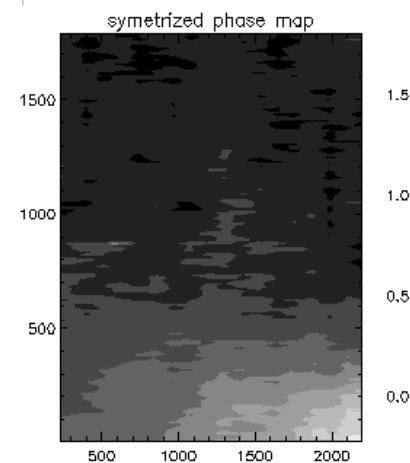
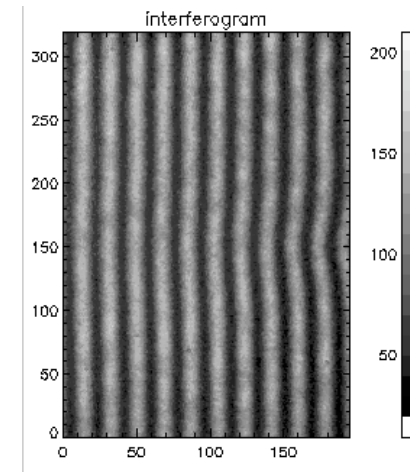
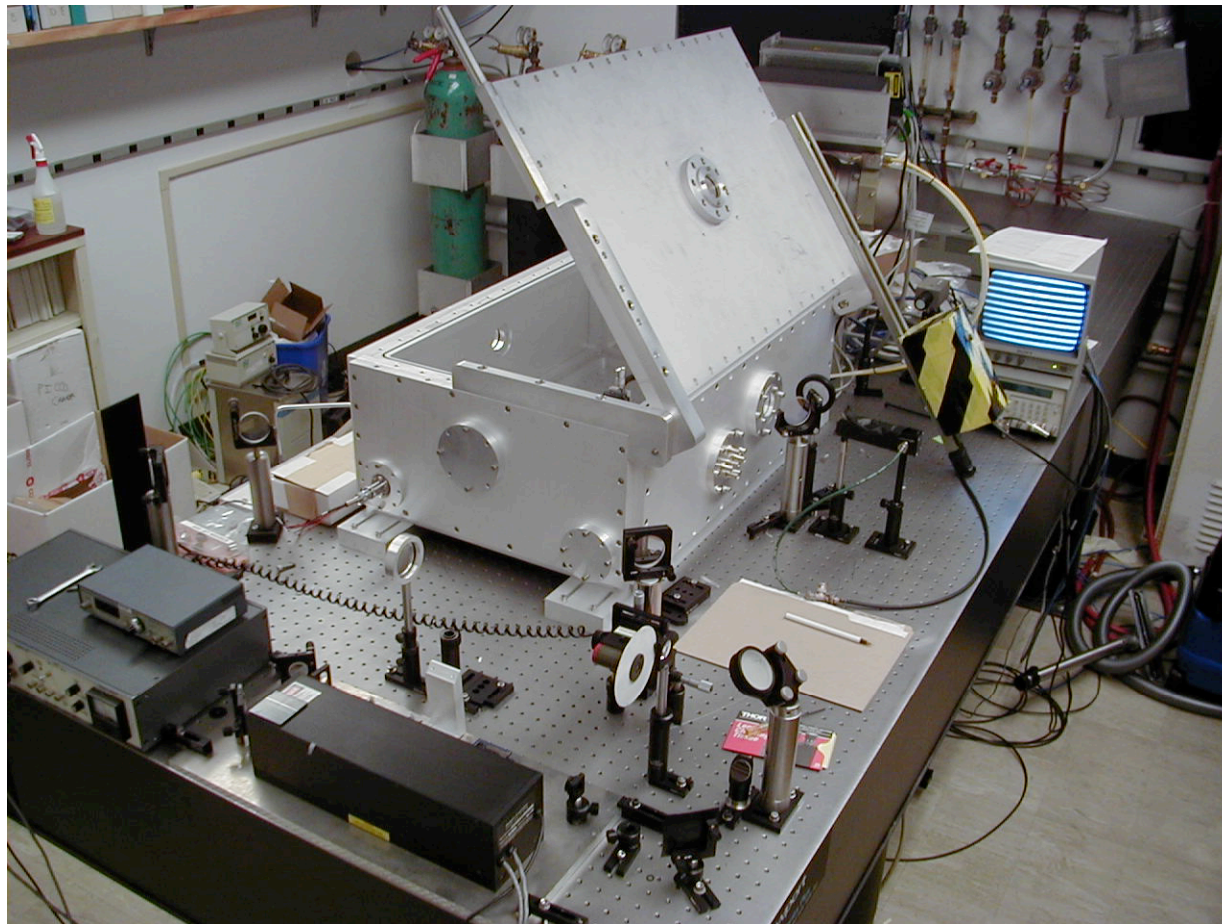
Slit Nozzle  $200\mu\text{m} \times 4\text{mm}$







## Test stand with HeNe neutral density interferometer allows fast characterization of nozzles

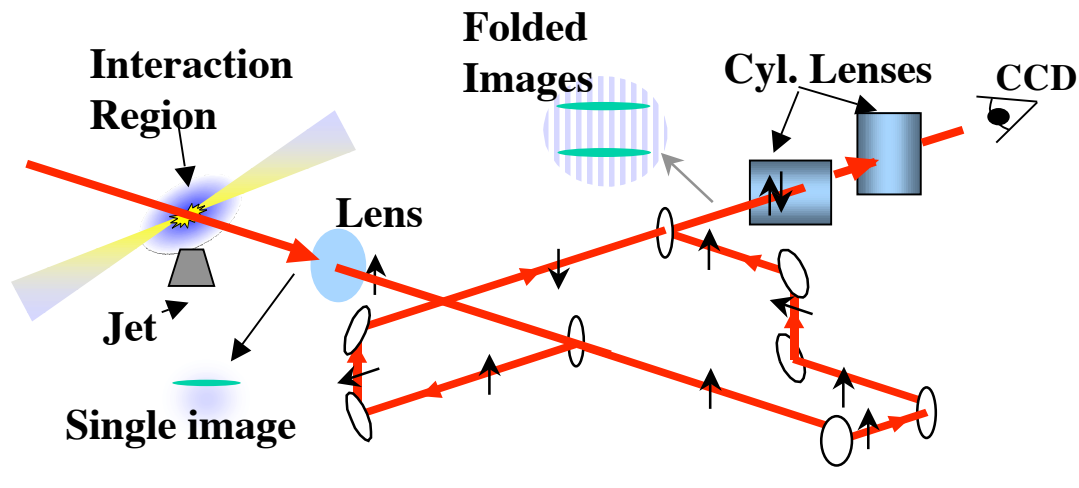


- Neutral density interferometer & fringe tracing recovers 2d phase map with  $\sim 0.05\text{rad}$  resolution
- Testing of cylindrical nozzles on solenoidal gas jets is under way
- Quick feedback allows rapid iteration of shapes



2 $\square$  interferometer with asymmetric imaging will measure plasma during system shots, separate sidescatter

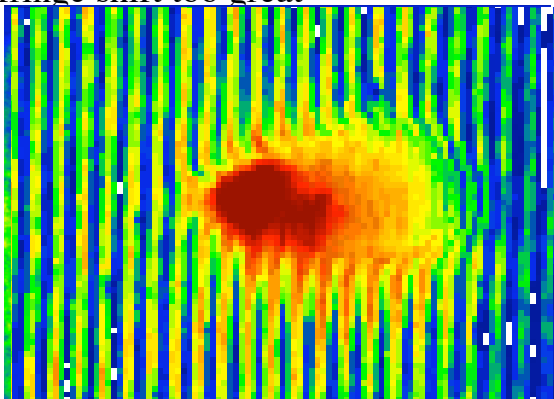
## Interferometer Setup



- Interferometer upgrade has been fielded on accelerator and is taking first data.
- 1.4mm \* 7mm asymmetric field of view allows imaging of long thin plasma channels
- Allows measurements of plasma at full system energy -> actual shot conditions

### 1 $\square$ Symmetric imaging :500 mJ

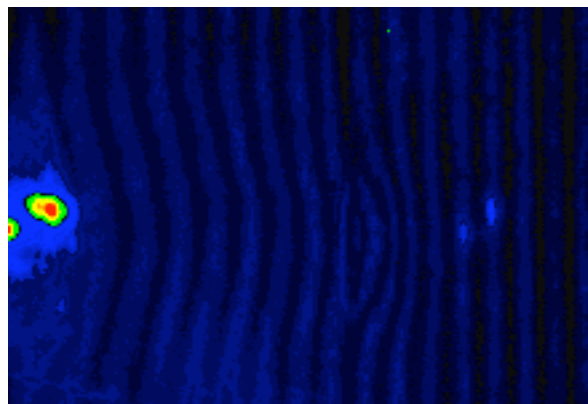
- sidescatter confuses interferogram-
- data un readable at high power
- insufficient resolution
- fringe shift too great



### First images: Asymmetric 2 $\square$ interferometer

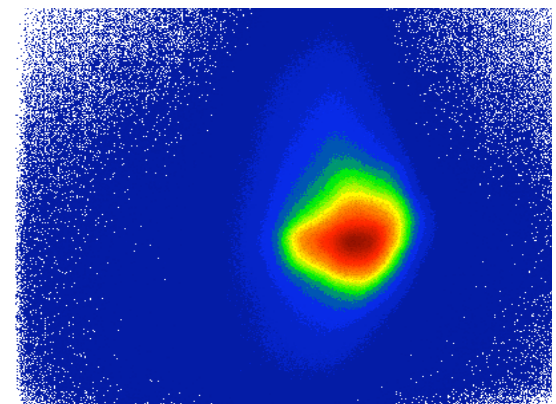
#### Partial interferogram: 500mJ

- sidescatter tolerable even at high power
- resolution, fringe shift good



#### Side Scatter isolated:50mJ

- Separate analysis possible

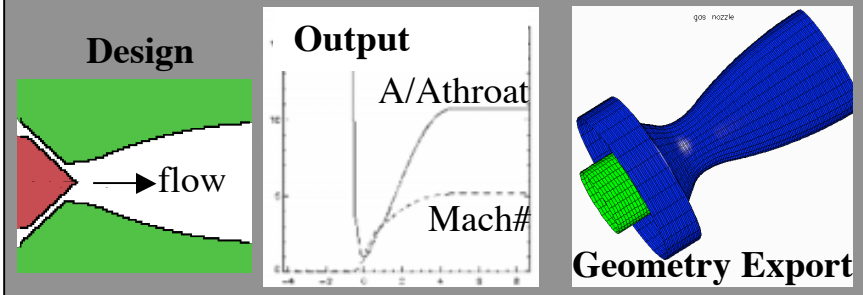






Conclusion: New gas nozzles/valves are being developed using  
Test stand interferometer, design/simulation software,  
and novel drivers

#### Dynamic Design Code & Interface to Simulations



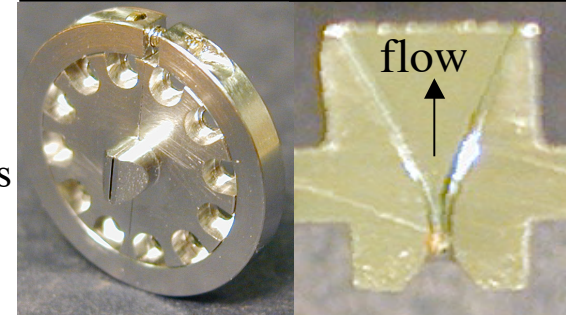
#### Design

- Quasi-1d gas dynamic design code allows fast nozzle design with feedback on mach profiles/etc.
- Geometry export interfaces to SciDAC flow simulations for 3d CFD nozzle simulations and to CMC machining codes

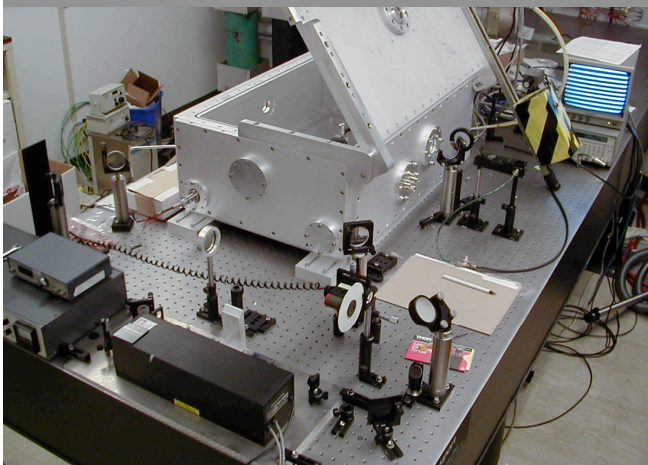
#### Production

- A PZT driven valve and driver have been designed and built to reduce opening time - starting testing
- Slit nozzle has been fabricated and is being tested to provide long, narrow jets
- Evaluating micro valve technologies for arrayed valves -> shaped gas profiles

#### Slit Nozzle 200 $\mu$ m x 4mm



#### Gas Jet Test Stand



#### Testing and Optimization

- HeNe interferometer with Chopper provides  $< 10\mu$ s time resolution
- Fringe tracing interferogram analysis software gives  $\sim 0.05$  rad phase resolution and allows recovery of density profiles from neutral gas or plasma interferograms
- Testing and optimization of cylindrical nozzles on solenoidal gas jets is under way
- Testing of slit nozzles and new drivers is beginning